

IN THE CLAIMS

Please amend claims as follows:

1. (Canceled) A manufacturing method for obtaining high-temperature components (10) for gas turbines, characterized in that it comprises at least incorporation of an element or insert (14) in a main body (12) of said component (10), said element or insert (14) having mechanical properties able to withstand stresses to which said component (10) is subject in a zone where the said element or insert (14) is arranged.

2. (Canceled) The manufacturing method according to claim 1, characterized in that wherein it envisages at least one fixing process for joining said element or insert (14) to said main body (12) of said component (10).

3. (Canceled) A high-temperature component (10) for gas turbines, characterized in that wherein it is obtained by incorporating at least one element or insert (14) in a main body (12) of said component (10), said element or insert (14) having mechanical properties able to withstand stresses to which said component (10) is subject in a zone where said element or insert (14) is arranged.

4. (Currently Amended) The high-temperature component Component (10) according to claim [[3]]10, characterized in that wherein said element or removable insert (14) is of the modular type.

5. (Currently Amended) The high-temperature component Component (10) according to claim [[3]]10 characterized in that wherein said element or removable insert (14) is arranged in an inlet zone of a blade.

6. (Currently Amended) The high-temperature component Component (10) according to claim [[3]]10, characterized in that wherein said element or removable insert (14) is arranged in an outlet zone of a blade.

7. (Currently Amended) The high-temperature component Component (10) according to claim 3, ~~characterized in that~~ wherein said element or removable insert (14) is made of a material which is more resistant to high temperatures than the material of said main body (12).

8. (New) A manufacturing method for obtaining high-temperature components for gas turbines, the method comprising:

producing a removable insert of a material that is more resistant to high temperatures than a material of a main body of a component of a gas turbine; and

fixing the removable insert to the main body of the component in a zone, wherein the removable insert is operable to have mechanical properties to withstand stresses to which the component is subjected in the zone, and wherein the removable insert is operable to be removed from the main body when the removable insert becomes damaged, while retaining the main body that is still fit for operation.

9. (New) The method of claim 8, further comprising replacing the removable insert with another removable insert when the removable insert becomes damaged, by fixing the additional removable insert to the zone of the main body.

10. (New) A component for gas turbines, comprising:

a main body; and

a removable insert configured to be fixed to a zone of the main body, wherein the removable insert is operable to have mechanical properties to withstand stresses to which the component is subjected to in the zone where the insert is arranged, and wherein the removable insert is operable to be removed from the main body when the removable insert becomes damaged, while retaining the main body that is still fit for operation.